

FULL-ORDER AND REDUCE-ORDER LINEAR SYSTEM WITH UNKNOWN INPUT IMPLEMENTATION IN MATLAB/SIMULINK

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ABSTRACT

In this paper full Full-Order and Reduced-Order Linear Systems with Unknown Inputs Implementations are test both full order and reduce order observer design and examination of the stability of any linear system. The system stability test by the coordinate system they system are used provided the all existence condition. The full order observer are observed in all state and reduce order observer is estimated by the state reduce order observer are more complex. The advantages of the reduce observer avoiding the reconstructing accessible states design.

Drawback and advantage the examination technique is pointed and illustrated by the matlab simulation in numerical examples. The theoretical results have also been developed how to set up the reduced-order initial conditions using the least-squares method, derived and an observation that the reduced-order observer output is identical to the original system's actual output, the result established by matlab/simulink model.

KEYWORDS: Matlab simulation, full order observer, reduce order observer, unknown input, linear system.

1. INTRODUCTION

The proposed system are Hopefully by understanding full- and reduced-order observer design and Matlab/Simulink implementation, students, instructors, engineers, and scientists will

appreciate the importance of observers and feel confident using observers and observer-based controllers in numerous engineering and scientific applications. In the control systems, time invariant linear systems are represented, in state-space form, by,

$$\frac{dx(t)}{dt} = Ax(t) + Bu(t), \quad x(0) = x_0 \quad (1)$$

Where $x(t)$ is the state-space vector of dimension n , $u(t)$ is the system input vector (which may be used as a system control input) of dimension m , and matrices A and B are constant and of appropriate dimensions. In practice, the initial condition is often unknown, in which case an observer is designed. estimate or observe system state-space variables at all times.

To take the advantage of the useful features of feedback (see, for example, it is often assumed that all state variables are available for feedback (full-state feedback), allowing that a feedback control input can be applied as,

$$u(x(t)) = -Fx(t), \quad (2)$$

Where F is a constant feedback matrix of dimension $m \times n$. The fact that all state-space variables must be available for feedback is a prevalent implementation difficulty of full-state feedback controllers. Moreover, large-scale systems with full-state feedback have many feedback loops, which might become very costly and/or impractical. Moreover, often not all state variables are available for feedback. Instead, an output signal that represents a linear combination of the state-space variables is available,

$$y(t) = Cx(t) \quad (3)$$

Where $\dim y(t) = l < n = \dim x(t)$. It is assumed that

$l = c = \text{rank } C$, so there are no redundant measurements. In such a case, under certain conditions, an observer can be designed that is a dynamic system driven by the system input and output signals with the goal of reconstructing (observing, estimating) all system state-space variables at all times.

This article shows how to implement full- and reduced-order observers using the Matlab and Simulink software packages for computer-aided control system design. In fact, how to implement a linear system and its observer, represented by their state-space forms, using the Simulink state-space blocks is shown in below,

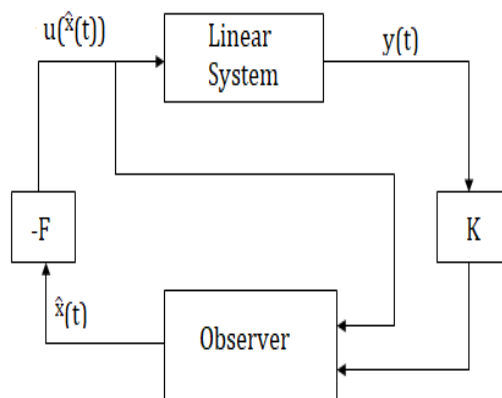


Fig. 1: A block diagram for the system observer configuration.

2. FULL-ORDER OBSERVER DESIGN

The theory of observers originated in the mid 1960s. According to any system driven by the output of the given system can serve as an observer for that system. Consider a linear dynamic system,

$$\begin{aligned} \dot{x}(t) &= Ax(t) + Bu(t), \quad x(0) = x_0 = \text{unknown}, \\ y(t) &= Cx(t). \end{aligned} \tag{4}$$

The system output variables $y(t)$ are available at all times, and that information can be used to construct an artificial dynamic system of the same order n^{th} as the system under consideration that can estimate the system state-space variables at all times. Since the matrices A, B, C are known, it is rational to postulate an observer for (4) as,

$$\begin{aligned} \dot{\hat{x}}(t) &= A\hat{x}(t) + Bu(t), \\ \hat{x}(t_0) &= \hat{x}_0, \end{aligned} \tag{5}$$

If the outputs $y(t)$ and $\hat{y}(t)$ are compared, they will, in general, be different since, in the first case, the initial condition of (4) is unknown, and, in the second case, the initial condition of the proposed observer (5) is chosen arbitrarily by a control engineer (designed). The difference between these two outputs generates an error signal,

$$y(t) - \hat{y}(t) = Cx(t) - C\hat{x}(t) = Ce(t), \tag{6}$$

Which can be used as the feedback signal to the observer such that the estimation (observation) error $e(t) = x(t) - \hat{x}(t)$ is reduced. An observer that takes into account feedback information about the observation error,

$$\begin{aligned} \dot{\hat{x}}(t) &= A\hat{x}(t) + Bu(t) + K(y(t) - \hat{y}(t)) \\ A\hat{x}(t) + Bu(t) + KCe(t), \end{aligned} \quad (7)$$

Where the matrix K represents the observer gain, which must be selected such that the observation error tends to zero as time increases. From (4) and (7), comes an expression for dynamics of the observation error,

$$\begin{aligned} \dot{e}(t) &= (A - KC)e(t), \\ e(t_0) &= \text{unknown}. \end{aligned} \quad (8)$$

If the observer gain K is chosen such that the feedback matrix $A - KC$ is asymptotically stable (has all eigenvalues with negative real parts), then the estimation error $e(t)$ will decay to zero for any initial condition $e(t_0)$. This stabilization requirement can be achieved if the pair (A, C) is observable. It can be also noted that the error (and observer) stabilization can be achieved under a weaker condition that the pair (A, C) is detectable (not all, but at least the unstable modes of matrix A are observable).

A standard rule of thumb is that an observer should be designed such that its response is much faster than the system. This is especially important when the observed state variables are used for the purpose of feedback control. This can be achieved theoretically by choosing the observer eigenvalues to be about ten times faster than the system eigenvalues. That is by setting the smallest real part of the observer eigenvalues to be ten times larger than the largest real part of the closed-loop system eigenvalues.

Since the system changes in time, its estimated state variables must be as current as possible, otherwise the feedback signals represent considerably delayed estimates of the actual state variables, which can make the controller inaccurate and inefficient.

$$\left| \operatorname{Re}\{\lambda_{\min}(A-KC)\} \right|_{\text{observer}} > 10 \left| \operatorname{Re}\{\lambda_{\max}(A-BF)\} \right|_{\text{system}} \quad (9)$$

Theoretically, an observer can be made arbitrarily fast by locating its closed-loop eigenvalues very far to the left in the complex plane, but very fast observers can generate noise, which is not desirable.

After the system eigenvalues λ_i ($A - BF$) are determined, the observer eigenvalues λ_i ($A - KC$) are placed in the desired locations by selecting the corresponding observer gain K using the eigenvalue assignment technique.

2.1 Separation Principle

It is important to point out that the system-observer configuration preserves the closed-loop system eigenvalues that would have been obtained if the linear, perfect-state-feedback control had been used. This fact is shown below.

The system (4) under perfect state-feedback control, that is, $u(x(t)) = -Fx(t)$ has the closed-loop form,

$$\dot{x}(t) = (A - BF)x(t), \quad (10)$$

So that the eigenvalues of the matrix $A - BF$ are the closed-loop system eigenvalues under perfect state feedback. In the case of the system-observer configuration, the actual control signal applied to both the system and the observer is

$$u(\hat{x}(t)) = -F\hat{x}(t) = -Fx(t) + Fe(t) \quad (11)$$

From (8), (10), and (11),

$$\begin{bmatrix} \dot{x}(t) \\ \dot{e}(t) \end{bmatrix} = \begin{bmatrix} A - BF & BF \\ 0 & A - KC \end{bmatrix} \begin{bmatrix} x(t) \\ e(t) \end{bmatrix} = A \begin{bmatrix} x(t) \\ e(t) \end{bmatrix}$$

Since the state matrix of this augmented system is upper block triangular, its eigenvalues are equal to the union of the eigenvalues of the matrices $A - BF$ and $A - KC$. A very simple relation among $x(t)$, $e(t)$, and $\hat{x}(t)$ can be written using the definition of the estimation error as

$$\begin{bmatrix} x(t) \\ e(t) \end{bmatrix} = \begin{bmatrix} I & 0 \\ I & -I \end{bmatrix} \begin{bmatrix} x(t) \\ \hat{x}(t) \end{bmatrix} = T \begin{bmatrix} x(t) \\ \hat{x}(t) \end{bmatrix}$$

Note that the matrix T is nonsingular. To go from x -coordinates to \hat{x} -coordinates.

3. Full-Order Observer Implementation in Simulink

An observer, being an artificial dynamic system of the same order as the original system, can be built by a control engineer using either capacitors and resistors (what electrical engineers do) or using masses, springs, and frictional elements or simply using a simulink modal that simulates and solves the corresponding differential equation, which is something that

anybody with a basic knowledge of control systems and differential equations can do, especially those familiar with Matlab and Simulink.

$$\begin{aligned}\dot{\hat{x}}(t) &= A\hat{x}(t) + Bu(t) + K(y(t) - \hat{y}(t)) \\ \hat{y}(t) &= c\hat{x}(t)\end{aligned}\quad (12)$$

In practice, the observer is implemented as a linear dynamic system driven by the original system input and output signals, that is, $u(t)$ and $y(t)$. Eliminating $y(t)$ from the observer equation (12) yields the observer form used in practical implementations,

$$\dot{\hat{x}}(t) = (A - KC)\hat{x}(t) + Bu(t) + Ky(t). \quad (13)$$

The corresponding block diagram of the system-observer configuration, also known as the observer-based controller configuration, is presented in Figure with the feedback loop closed using the feedback controller as defined in namely $u(\hat{x}(t)) = -F\hat{x}(t)$.

The state-space block of Simulink is used. That block allows only one input vector and one output vector. For that reason, the observer is represented as a system with one augmented unknown input,

$$\dot{\hat{x}}(t) = (A - KC)\hat{x}(t) + [BK] \begin{bmatrix} u(t) \\ y(t) \end{bmatrix} \quad (14)$$

Using the given dimensions of the state, input, and output variables (respectively n , m , and c), the system matrices for the state-space Simulink blocks are set as follows. For the system, the matrices are respectively: A , B , C , $\text{zeros}(c, m)$, the last one meaning a zero matrix of dimensions cm .

The results obtained for $y(t)$ and $\hat{y}(t)$ and output observation error $e(t)$ may be presented using the Simulink block "scope," or passed to the Matlab window, together with time, where they can be plotted using the Matlab plot function. For example, `plot(t,error)` plots the output observation error as a function of time. t Such a plot (or figure) can be further edited using numerous Matlab figure-editing functions. For the observer, the corresponding matrices should be set using information from,

$$\begin{aligned}&\text{Observer State Space} \\ &\text{Block Setup} \\ &A - K * C, [B \ K], \text{eye}(n), \\ &\text{Zeros}(n, m+c), X_0\text{hat}\end{aligned}\quad (15)$$

The first matrix is the observer feedback matrix, the second is the augmented matrix of two inputs into the observer and the third matrix (set to the identity of n dimension) indicates that estimates of all n state variables are available on the observer output. Since the observer is a system either built by a designer or a computer program that simulates observer dynamics, the designer has full freedom to choose the observer output matrix and set it to an identity matrix so that all observed (estimated) state space variables appear on the observer output. The fourth matrix represents the matrix “D” in the observer block and, due to the fact that the input matrix into the observer is the augmented matrix $[B \ K]$ of dimension $n \times (m+c)$ and the output matrix from the observer is of dimension $n \times n$ (identity matrix I_n), the dimension of the zero matrix must be $n \times (m+c)$. Finally, the last entry in denotes the observer initial condition vector that can be set arbitrarily. Later on in the article, a rational choice of the observer initial condition will be discussed. These matrices can be entered in the Simulink state-space block by double left clicking on the observer state-space block, which will open a new window as shown in Figure For details, see “Matlab Code: System State-Space Block Setup.”

4. REDUCE-ORDER OBSERVER DESIGN

Consider the linear dynamic system defined with the corresponding measurements. Assume that the output matrix C has full rank (equal to c) so that there are no redundant measurements. This means that the output equation $y(t) = Cx(t)$ represents, at any time, c linearly independent algebraic equations for n unknown state variables $x(t)$. Note that $y(t)$, of dimension, c is the measured system output, and hence is known at all times. In this section, it is shown how to construct an observer of reduced order $r = n - c$ for estimating the remaining r state-space variables [c state variables can be obtained directly from the system measurements $y(t) = Cx(t)$]. As indicated the reduced-order observer should be used when the system measurements have no noise. If noise is present, it is better to use the full-order observer, since it filters the system measurements and, in general, all state variables.

It will be seen that the procedure for obtaining the reduced-order observer is not unique. An arbitrary matrix C_1 of dimension $r \times n$ whose rank is equal to $r = n - c$ can be found such that the augmented matrix,

$$\text{rank} \begin{bmatrix} C \\ C_1 \end{bmatrix} = n,$$

has full rank equal to n . Introduce a vector $p(t)$ of dimension r such that

$$p(t) = C_1 x(t)$$

then

$$\begin{bmatrix} y(t) \\ p(t) \end{bmatrix} = \begin{bmatrix} C \\ C_1 \end{bmatrix} x(t),$$

which can be solved to obtain

$$\hat{x}(t) = Ly(t) + L_1 \hat{p}(t)$$

Where $y(t)$ are known system measurements and $\hat{p}(t)$ has to be obtained using a reduced-order observer of dimension. $r < n$ As a preliminary result in the process of constructing a reduced-order observer for $p(t)$ it can be noticed from that the following algebraic relations exist,

$$I_n = \begin{bmatrix} C \\ C_1 \end{bmatrix} [L \ L_1] = \begin{bmatrix} CL & CL_1 \\ C_1L & C_1L_1 \end{bmatrix} = \begin{bmatrix} I_c & 0 \\ 0 & I_r \end{bmatrix} \quad (16)$$

giving $CL = I_c$, $C_L L_1 = I_r$, $CL_1 = 0$, $C_1L = 0$, where I_n , I_c , I_r are identity matrices of corresponding dimensions and $n = c + r$.

Since from $p(t) = C_1 x(t)$ the differential equation for $p(t)$ can be easily constructed using the original system differential equation which leads to,

$$\begin{aligned} \dot{p} &= \frac{d}{dt}(p(t)) = \frac{d}{dt}(C_1 x(t)) = C_1 \dot{x}(t) \\ &= C_1 Ax(t) + C_1 Bu(t) \\ &= C_1 AL_1 p(t) + C_1 Bu(t) \end{aligned} \quad (17)$$

To design an observer for $p(t)$ using the above established principal

$$\begin{aligned} Y(t) &= Cx(t) = CLy(t) + CL_1 p(t) \\ &= y(t) + 0 = y(t) \end{aligned} \quad (18)$$

More information about $y(t)$ can be obtained from the knowledge of $\dot{y}(t)$, in which case

$$\begin{aligned} \dot{y}(t) &= C\dot{x}(t) = CAx(t) + CBu(t) \\ &= CAL_1 p(t) + CALy(t) + CBu(t). \end{aligned} \quad (19)$$

This indicates that $\dot{y}(t)$ contains information about $p(t)$, so that it can be used to construct an observer for $p(t)$

$$\begin{aligned} \dot{\hat{p}}(t) &= C_1 AL_1 \hat{p}(t) + C_1 ALy(t) + C_1 Bu(t) + \\ &K_1 (\dot{y}(t) - \dot{\hat{y}}(t)), \end{aligned} \quad (20)$$

Where K_1 is the reduced-order observer gain, which has to be determined such that the reduced-order observer error $p(t) - \hat{p}(t)$ goes to zero as time increases. The reduced-order

observer measurement is obtained from by replacing $p(t)$ by its estimate (observation) $\hat{p}(t)$, that is,

$$\hat{y}(t) = CAL_1\hat{p}(t) + CALy(t) + CBu(t). \quad (21)$$

Signal differentiation is not a recommended operation since it is very sensitive to noise; signal differentiation amplifies noise so it should be avoided in all practical applications. Using an appropriate change of variables, it is possible to completely eliminate the need for information about the derivative of the system measurements $\dot{y}(t)$.

4.1 Setting Reduced-Order-Observer Eigenvalues in the Desired Locations

As discussed in good performance typically requires placing the reduced-order observer eigen values such that the reduced-order observer is roughly ten times faster than the system whose speed is determined by the closed loop system eigen values given by $\lambda(A-BF)$ Note that the reduced-order observer matrix can be written as,

$$A_q = C_1AL_1 - K_1CAL_1 = (C_1AL_1) - K_1(CAL_1)$$

To determine the reduced-order observer gain K_1 such that it arbitrarily places the reduced-order observer eigen values, the pair (C_1AL_1, CAL_1) must be observable. The next subsection will show that this condition is satisfied if the original system is observable (the pair (A, C) is observable). See “Reduced-Order-Observer Design Process” for details on the steps involved in the design of this observer.

4.2 Reduced-Order Observer Design with a Change of State Coordinates

It is shown first in this subsection that a reduced-order observer can be also designed with a change of state coordinates. More importantly, with this change of state coordinates, it becomes easy to prove the result that if the original system is observable then the reduced-order observer is also observable.

Consider a linear system with the corresponding measurements defined in (4). The reduced-order observer design can be simplified, and some important conclusions can easily be made using a nonsingular transformation $p^{n \times n}$

$$Px(t) = \begin{bmatrix} x_1(t) \\ x_2(t) \end{bmatrix}$$

Such that

$$\begin{aligned} Y(t) &= CX(t) = CP^{-1}PX(t) = CP^{-1} \begin{bmatrix} x_1(t) \\ x_2(t) \end{bmatrix} \\ &= [I_c \ 0] \begin{bmatrix} x_1(t) \\ x_2(t) \end{bmatrix} = x_1(t) \end{aligned} \quad (22)$$

To map the system into new coordinates in which the system measurements provide complete information about a part of the state-space vector $x_1(t)$ of dimension c . In such a case, an observer is needed to estimate the remaining part of the state-space vector, that is, $x_2(t)$ of dimension $r = n-c$. This is possible since there exists a transformation P of the full-rank matrix, C obtained using elementary transformations, such that $CP^{-1} = [I_c \ 0]$. This procedure can be found in many standard linear algebra. Such a transformation is defined in so that.

$$CP^{-1} = C[L \ L_1] = [CL \ CL_1] = [I \ 0]$$

5. Extension to Kalman Filtering

By mastering the design of deterministic observers presented in this article, more complex linear dynamic estimation problems can also be considered, for example, the Kalman filtering and other optimal linear estimation problem. In fact, the Kalman filter can also be implemented as a software program in Matlab/Simulink using the methodology presented in this article. For a linear dynamic system disturbed by a Gaussian white noise stochastic process $w(t)$ and system measurements corrupted by a Gaussian white noise stochastic process $v(t)$

$$\begin{aligned} \dot{x}(t) &= AX(t) + Bu(t) + Gw(t), \\ Y(t) &= CX(t) + v(t) \end{aligned} \quad (23)$$

The kalman filter is given by

$$\dot{\hat{x}}_{KF}(t) = (A-KC)\hat{x}_{KF}(t) + Bu(t) + K_{KF}Y(t), \quad (24)$$

Which is exactly the same structure as the full-order observer structure is presented. The Kalman filter gain K_{KF} and the Kalman filter initial conditions must be chosen using results from optimal Kalman filtering theory.

6. Full And Reduce Order Observer Designs For An Aircraft Model

$$A = \begin{bmatrix} -0.01357 & -32.2 & -46.3 & 0 \\ 0.00012 & 0 & 1.214 & 0 \\ -0.0001212 & 0 & -1.214 & 1 \\ 0.00057 & 0 & -9.1 & -0.6696 \end{bmatrix}$$

$$B = \begin{bmatrix} -0.433 \\ 0.1394 \\ -0.1394 \\ -0.1577 \end{bmatrix}$$

$$C = \begin{bmatrix} 0 & 0 & 0 & 1 \\ 1 & 0 & 0 & 0 \end{bmatrix}$$

$$D = \begin{bmatrix} 0 \\ 0 \end{bmatrix}$$

Matrix C_1 needed for the reduce-order observer design is chosen

$$C_1 = \begin{bmatrix} 0 & 1 & 0 & 0 \\ 0 & 0 & 1 & 0 \end{bmatrix}$$

See “Matlab Program for the Aircraft Simulation Example” for the corresponding Matlab program with all simulation data. The obtained differences between the actual and estimated state trajectories using the full- and reduced-order observers are presented in Figure 2 and 3. In both cases, the initial conditions for the observers are obtained using the least-squares formulas. It can be seen from Figures 4 to 7 that the reduced-order observer performs better than the full-order observer. Not only are there no observation errors for the two state variables directly measured, but the reduced-order observer is also more accurate than the full-order observer. Note that the eigenvalues for both observers are placed to be of the same speed.

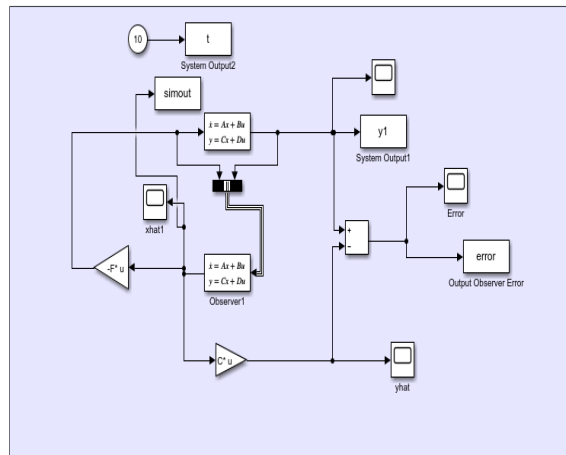
The reduced-order observer eigenvalues are placed at -10, -11 and the full-order observer eigenvalues are placed at -10, -11, -12, -13. Moreover, the reduced-order observer is simpler for implementation, since it is a dynamic system of lower order than the original system. An additional advantage of the reduced-order observer is that they require fewer sensors and fewer feedback loops for corresponding feedback control applications.

The importance of Matlab/Simulink implementation of the full- and reduced-order observers has been demonstrated in the previous sections of this article. These results can be used in all areas of science and engineering.

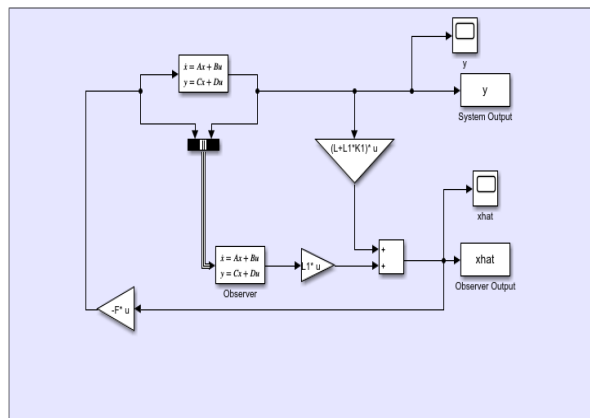
7. SIMULATION AND RESULT

The Full-Order and Reduce-Order observer is obtained by matlab simulink the system is work on the pole placement method it is find the system stability. On the system stable then method is not work but the system are unstable then the find unstable pole location and pole is place by desired location in required characteristic equation on the system the simulink modal and simulink result in shown below.

Full Order Observer



Reduced Order Observer



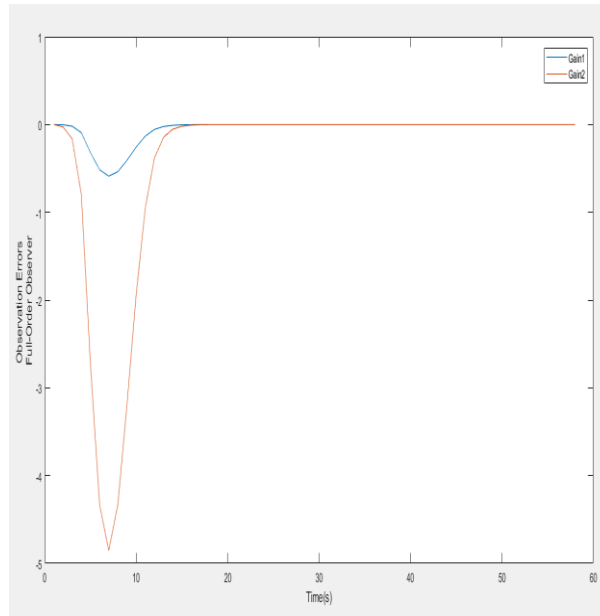


Fig. 4: System error.

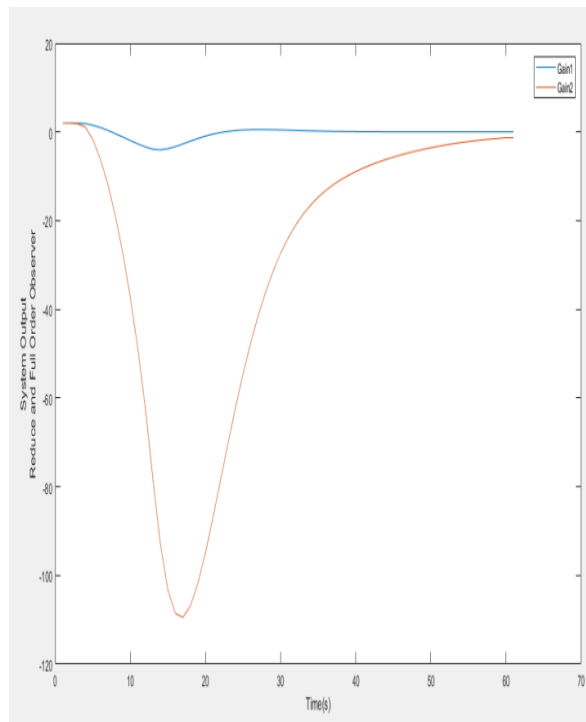


Fig. 5: System output of reduce and full order observer.

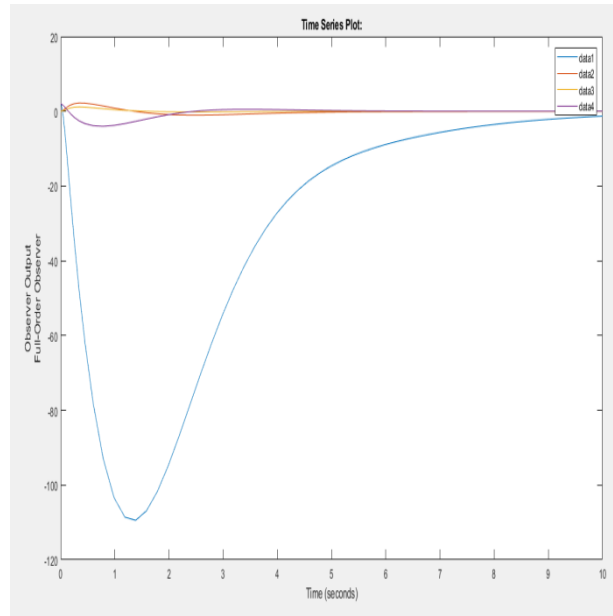


Fig. 6: Full-order Output.

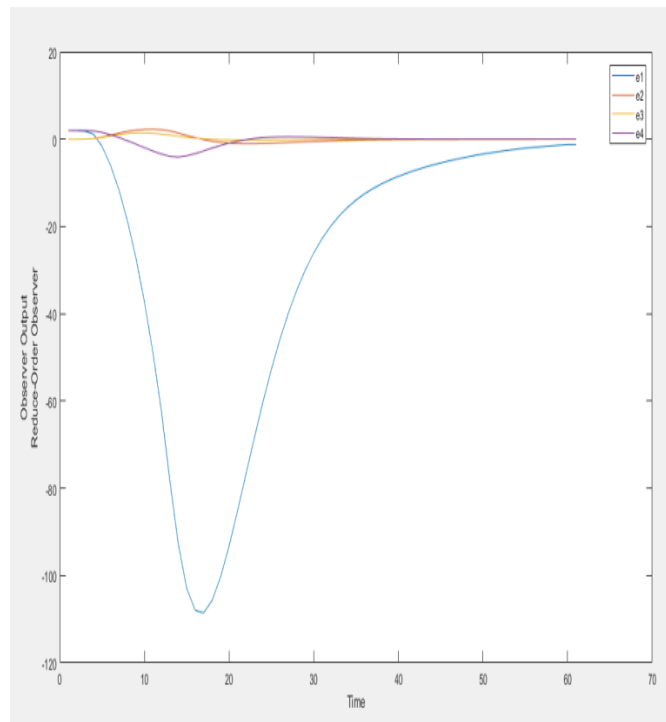


Fig. 7: Reduce-order Output.

8. CONCLUSION

In this proposed system has been shown in detail how to implement Full-order and Reduce-order observer in matlab/simulink. The fundamental method are conclude by the system stability.

The stability are find in the drown in the figure by given characteristic equation to check the stability of any system. In this method are implement by the required input and output are increase then the system are use in the more input and also more output in future work.

The system are use in the closed loop system it main work to get required output in the system. The system are more useful in the find system stability it is very easy to implement in the personal and industrial uses to change the system input and output.

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